

Order Reduction of MIMO systems using Firefly Algorithm

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ABSTRACT

A new model reduction technique of linear multi variable system is proposed using the prominence of Firefly Algorithm (FA). The denominator polynomial of the lower order transfer function matrix is determined by Dominant pole retention method by retaining the dominant poles of original stable higher order system .While the numerator polynomial of the lower order transfer function matrix is determined by minimizing Integral Square Error (ISE) between the transient responses of original and lower order models using FA. The efficacy of the algorithm is tested through a 10th order model of a Single Machine Infinite Bus (SMIB) power system.

Keywords: Dominant pole retention method, Firefly Algorithm, Integral Square Error, Lower Order Model, Multi variable system.

1. INTRODUCTION

Analysis of a large interconnected power system is extremely time consuming and may even exceed the storage capacity of modern fast computers because of high order system matrix. The complexity often makes it difficult to obtain a good understanding behavior of a system. The proper analysis of high order systems is very tedious and costly. If a low order linear model is derived for a high order system then the preliminary design and optimization is achieved with very much ease. Hence, methods have to be developed to obtain simplified models of the systems by using different reduction techniques. Several methods are available in international literature for the model reduction of high-order systems such as dominant pole [1], Pade Approximation [2], Stability Equation method [3], and Routh Approximation method [4]. To improve the correlation between original higher order and lower order model some of the mixed methods were developed, where the denominator polynomial is determined by preserving the stability of higher order system while the numerator polynomial is obtained using mixed method with easy numerical implementation such as Routh-Pade [5], Dominant pole-Pade [6], Stability equation-Pade [7]. All these methods have their own merits and demerits when used on a particular system.

A large number of order reduction methods of multivariable systems based on step error minimization are also developed [8][9]. In these methods, the denominator polynomial of Lower Order Model (LOM) is chosen so as to preserve stability of higher order system such as dominant pole, Routh approximation methods, etc. and then the numerator polynomial of the LOM are determined by minimization of the ISE [10] to determine the transient responses of original and lower order models to obtain optimal value. Usage of optimization methods became familiar in most of applications in various disciplines. Several optimization techniques are proposed based on inspiration of nature known as evolutionary techniques. Genetic Algorithm (GA) [11], Particle Swarm Optimization (PSO)[12] etc., are some of the techniques based artificial intelligence. GA is based on survival of the fittest with crossover and mutation operations. PSO is based on bird flocking or fish schooling, which is similar to GA in initialization of population of random solutions and updating values to get optimal solution. PSO is popular as it has no evolution operators. One of the most promising advantage of PSO over GA is its simplicity, as it uses a few parameters and easy to implement.

In spite of various existing optimization methods, there is greater need for global optimization methods for MIMO systems. One of the most recent developments is based on inspiration of behavior of fireflies called Firefly algorithm [13], having few similarities of the most prominent algorithm PSO, which has been successfully applied to various applications in power systems [14-17], digital image processing [18] etc., and proved its fast convergence, easy implementation and wide exploration of search space than most of the existing popular techniques. The proposed method is based on developing a lower order model of a stable higher order MIMO system using firefly algorithm and Dominant pole retention method. Paper is organized as follows: In Section 2, nature inspired Firefly Algorithm is explained. In

Section 3, problem formulation. In Section 4, Comparison of results with other methods available in literature.

2. FIREFLY ALGORITHM (FA)

The Firefly Algorithm (FA) is a recently developed nature inspired algorithm by Xin-She Yang [13]. Based on the flashing of fireflies in the summer in the tropical temperature regions. Fireflies are of different species, to attract a prey each of them produces their own pattern. Based on intensity of brightness it can also communicate to other fireflies. This communication decreases as the distance between fireflies increases. The behavior of fireflies such as their attractiveness and communication leads to the inspiration for FA. The main advantage of Firefly is its fast convergence with global communication among all the fireflies, it is more effective in various optimization problems. The flashing of fireflies is associated with objective function to solve more optimization problems. For simplicity in describing our new FA the following three idealized rules are considered.

- All fireflies are unisex, and they will move towards more attractive and brighter ones.
- Degree of attractiveness of a firefly is proportional to its brightness.
- Brightness of a firefly is determined by the value of the objective function of a given problem.

In FA method, assume that there exists a swarm of pop (fireflies) solving the order reduction problem iteratively and $X(i,k)$ represents a solution for a firefly 'i' in algorithm iteration k. Initially all fireflies are in random manner. Each firefly has its own attractiveness which shows how strong it attracts other members of the swarm. Firefly attractiveness is determined by

$$\beta = \beta_0 e^{-\gamma r_{ij}^2} \quad \dots \quad 2.1$$

where r_{ij} is Cartesian distance between two fireflies, β_0 and γ are maximum attractiveness and absorption coefficient values respectively which are input parameters to this method. To explore search space effectively firefly i is changing its position iteratively based on attractiveness of other swarm members with higher light intensity i.e. $i_j > i_i$; for all $j = 1 \dots m$ and $j \neq i$ which is varying across distance and a fixed random step vector u_i for each search space dimension k is

$$u(i, k) = \alpha * (rand - 0.5) \quad \dots \quad 2.2$$

Where α being the randomization parameter and rand is a random number between 0 and 1.

The movement of a firefly i, attracted to another brighter firefly j, is expressed as

$$\mathbf{X}(i,k)=\mathbf{X}(i,k)+\beta_0*\exp(-\gamma r_{ij}^2*(x_j-x_i)+\alpha*(\text{rand}-0.5) \quad \dots \quad 2.3$$

terminate the method if the desired solution is obtained or maximum no. of iterations are reached.

b. FLOWCHART:

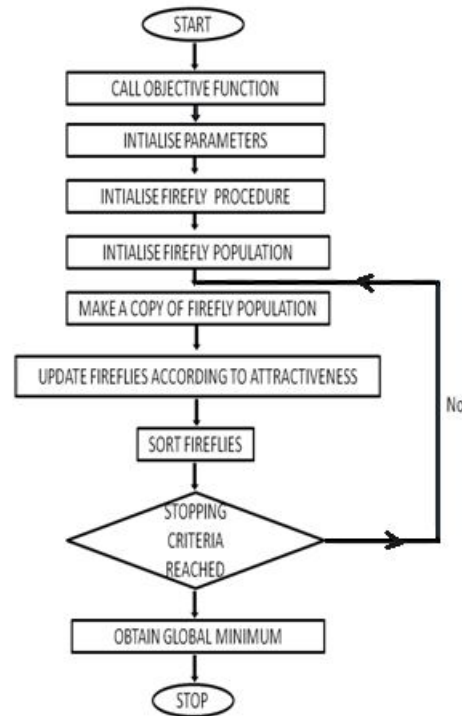


Fig 2.1 Flowchart of Firefly Algorithm

3. PROBLEM FORMULATION

A synchronous machine supplying power through a step-up transformer and a high-voltage transmission line to an infinite grid is considered (SMIB) as shown in Fig 3.1. In this system under study, X_t and X_l represents the reactance of the transformer and the transmission line respectively; V_t and E_b are the generator terminal and infinite bus voltage, respectively.

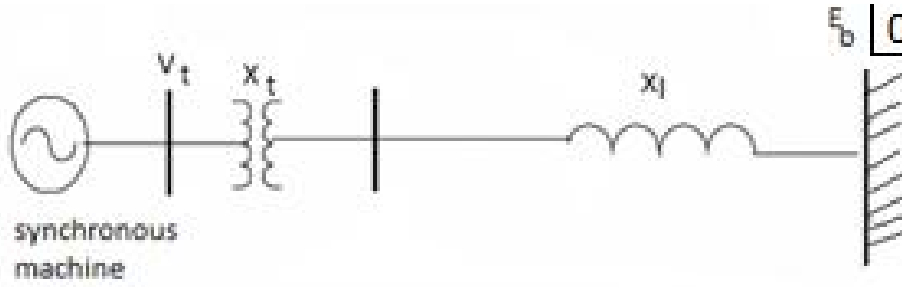


Fig 3.1 A simple SMIB power system

The linearised model of system consisting of a three-phase synchronous machine with a standard IEEE Type-I exciter with Rate Feedback (RF) and Power System Stabilizer (PSS) as shown in Fig 3.2 is expressed in terms of Heffron and Phillips constants.

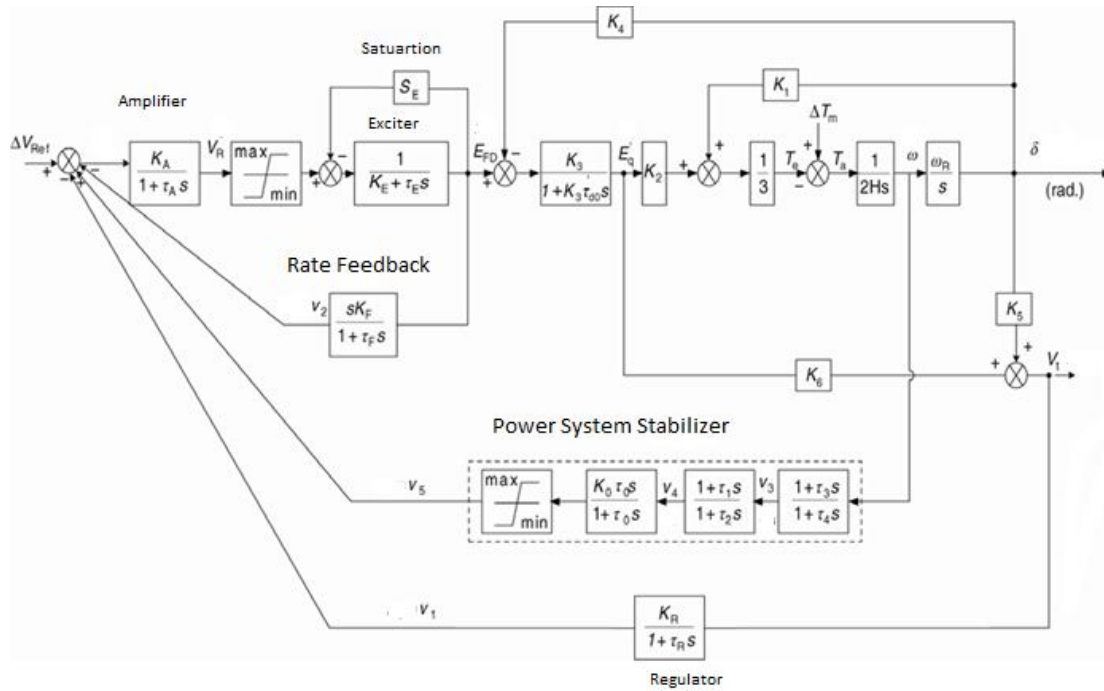


Fig 3.2 Block diagram representation of Heffron- Phillips model of SMIB power system

In general a model can be mathematically represented in state-space form as:

$$\dot{X} = AX + BU \quad \text{and} \quad Y = CX + DU \quad \dots \quad 3.1$$

$$X^T = [E_q' \omega \delta V_1 V_2 V_3 V_4 V_5 V_R E_{FD}] \text{ (State -variables)}$$

$$U^T = [\Delta V_{Ref} \Delta T_m] \text{ (Inputs)}$$

$Y^T = [\delta V_t]$ (Outputs) with their numerical values and operating point of the system in Appendix-A.

Based on the numerical values of parameters and operating point of system, required matrixes can be obtained.

An n^{th} order multi-input multi-output (MIMO) linear time invariant Higher Order System (HOS) in general can be described by a transfer matrix with 'j' inputs and 'i' outputs.

$$[G(s)] = \frac{1}{D(s)} \begin{bmatrix} b_{11} & b_{12} & \cdots & \cdots & b_{1j} \\ b_{21} & b_{22} & \cdots & \cdots & b_{2j} \\ \vdots & \vdots & \cdots & \cdots & \cdots \\ \vdots & \vdots & \cdots & \cdots & \cdots \\ b_{i1} & b_{i2} & \cdots & \cdots & b_{ij} \end{bmatrix} = [G_{kl}(s)] \quad \dots \quad 3.2$$

where $k=1,2,\dots,i$ and $l=1,2,\dots,j$. The general form of $[G_{kl}]$ is considered as

$$G_{kl}(S) = \frac{b_{kl}(s)}{D(s)} = \frac{b_n s^{n-1} + \dots + b_1}{a_n s^n + a_{n-1} s^{n-1} + \dots + a_0} \quad \dots \quad 3.3$$

Where $b_{kl}(s)$ and $D(s)$ are the different numerator polynomials in transfer function matrix and denominator polynomial of the HOS which are the Laplace transforms of the output variable $b_{kl}(t)$ and the input variable $D(t)$ respectively. It is required to obtain the m^{th} ($m < n$) LOM and it is defined as:

$$[r(s)] = \frac{1}{\bar{D}(s)} \begin{bmatrix} d_{11} & d_{12} & \cdots & \cdots & d_{1j} \\ d_{21} & d_{22} & \cdots & \cdots & d_{2j} \\ \vdots & \vdots & \cdots & \cdots & \cdots \\ \vdots & \vdots & \cdots & \cdots & \cdots \\ d_{i1} & d_{i2} & \cdots & \cdots & d_{ij} \end{bmatrix} = [r_{kl}(s)] \quad \dots \quad 3.4$$

In general form $[r_{kl}(s)]$ is considered as

$$r_{kl}(s) = \frac{d_{kl}(s)}{\bar{D}(s)} = \frac{d_m s^{m-1} + \dots + d_1}{c_m s^m + c_{m-1} s^{m-1} + \dots + c_0} \quad \dots \quad 3.5$$

Dominant pole retention method is one of the most familiar order reduction techniques available in literature. It provides a stable lower order model provided the Original higher order system is stable. This method has been mixed with several methods to improve its effectiveness. The denominator polynomial of lower order transfer function is obtained by using Dominant pole retention method

Consider $D(s)$ which is denominator polynomial of HOS

$$D(s) = a_n s^n + a_{n-1} s^{n-1} \dots + a_0 \quad \dots \quad 3.6$$

Compute the poles of $D(s)$ and retain the most dominant poles based on the required system dynamic performance.

Formulating the reduced order denominator polynomial $\tilde{D}(s)$ by retaining poles of original HOS.

Numerator polynomials of proposed method are determined using FA by minimizing step ISE between HOS and LOM.

$$\min J(d_1, \dots, d_{m-1}) = \int_0^{\infty} [e_m(t) - e(t)]^2 dt \quad \dots \quad 3.7$$

Where $d_1 \dots d_{m-1}$ are the numerator coefficients of lower order transfer function matrix, $e_m(t)$ the unit step response of the given lower order system at time 't' and $e(t)$ is the unit step response of the higher order system while also satisfying the steady state constraint.

To match the steady state values between the HOS and LOM.

$$d_1 = \frac{b_1 * c_0}{a_0} \quad \dots \quad 3.8$$

Using the recursive algorithm introduced by Astrom [10], the integral in equation 3.7 is determined in terms of γ_i and δ_i co-efficients of error function

$$R(s) = e_m(s) - e(s) = \frac{1}{s} (r_{kl}(s) - G_{kl}(s)) \quad \dots \quad 3.9$$

Thus, required LOM is obtained from the original HOS by minimizing ISE between them with no steady state response error.

4. RESULTS AND ANALYSIS

This linear time invariant multivariable 10th order practical system [12] under study is given by

The transfer function matrix of the power system under study is given by

$$[G(s)] = \frac{1}{D(s)} \begin{bmatrix} b_{11}(s) & b_{12}(s) \\ b_{21}(s) & b_{22}(s) \end{bmatrix} \quad \dots \quad 4.1$$

Where

$$D(s) = s^{10} + 64.21 s^9 + 1596 s^8 + 1.947e004 s^7 + 1.268e005 s^6 + 5.034e005 s^5 + 1.568e006 s^4 + 3.236e006 s^3 + 4.055e006 s^2 + 2.902e006 s + 2.528e005.$$

Poles of $D(s)$ are -19.0451 + i2.4859, -19.0451 - i2.4859, -11.9632, -9.6454, -0.2392 + i3.2348, -0.2392 - i3.2348, -2.1375, -0.8977 + i1.3552, -0.8977 - i1.3552, -0.1001.

and

$$b_{11}(s) = -2300 s^5 - 9.853e004 s^4 - 1.378e006 s^3 - 6.843e006 s^2 - 6.105e006 s - 5.435e005.$$

$$b_{12}(s) = 29.09 s^8 + 1868 s^7 + 4.609e004 s^6 + 5.459e005 s^5 + 3.185e006 s^4 + 8.701e006 s^3 + 1.206e007 s^2 + 7.603e006 s + 6.481e005.$$

$$b_{21}(s) = 85.23 s^7 + 3651 s^6 + 5.208e004 s^5 + 2.98e005 s^4 + 8.467e005 s^3 + 3.102e006 s^2 + 2.75e006 s + 2.448e005.$$

$$b_{22}(s) = -1.26 s^8 - 85.17 s^7 - 2089 s^6 - 2.568e004 s^5 - 1.909e005 s^4 - 7.122e005 s^3 - 1.084e006 s^2 - 2.966e005 s - 1.936e004.$$

The LOMs are obtained for the above linear multi variable HOS by using proposed algorithm.

Denominator polynomial obtained using Dominant pole retention Method is

$$\tilde{D}(s) = s^3 + 0.5785s^2 + 10.5690 s + 1.0532$$

Poles -0.1001 , $-0.2392 - i3.2348$ and $-0.2392 + i3.2348$ of HOS are retained in order mimic dynamic characteristics of the original HOS. As the retained poles are lying on left half of S-plane represents it is a stable lower order denominator obtained form a stable HOS.

Numerator polynomials of LOM are obtained by FA. Performance of the algorithm depends on input parameters which should be considered carefully to achieve the best optimal value. Parameters $\alpha = 0.95$, $\gamma = 0.8$, $\beta_0 = 1.0$, population size = 20 and maximum no. of iterations 500 are considered in the proposed algorithm and implemented in MATLAB.

The general form of 3rd order lower transfer function matrix is

$$[r(s)] = \frac{1}{\tilde{D}(s)} \begin{bmatrix} d_{11}(s) & d_{12}(s) \\ d_{21}(s) & d_{22}(s) \end{bmatrix} \quad \dots \quad 4.2$$

where

$$\tilde{D}(s) = s^3 + 0.5785s^2 + 10.5690 s + 1.0532 \text{ and}$$

$$d_{11}(s) = 3.1645s^2 - 16.6006s - 2.2595$$

$$d_{12}(s) = -0.8752s^2 + 29.2024s + 2.6977$$

$$d_{21}(s) = -0.6040 s^2 + 8.0311s + 1.0195$$

$$d_{22}(s) = 0.5170 s^2 - 1.5679s - 0.0808$$

The resonabilness of the 3rd order lower models acquired above is assessed by measuring similarity between the time responses of outputs of the original HOS and LOM, concerned to the same input step variation. These responses are shown in Fig 4.1-4.6 which are also compared with lower order models obtained by [12], with three definite input changes.

- When $\Delta V_{\text{Ref}}(s) = 0.05$ p.u. and $\Delta T_m(s) = 0$.
- When $\Delta V_{\text{Ref}}(s) = 0$ and $\Delta T_m(s) = 0.05$ p.u.
- When $\Delta V_{\text{Ref}}(s) = 0.05$ p.u and $\Delta T_m(s) = 0.05$ p.u.

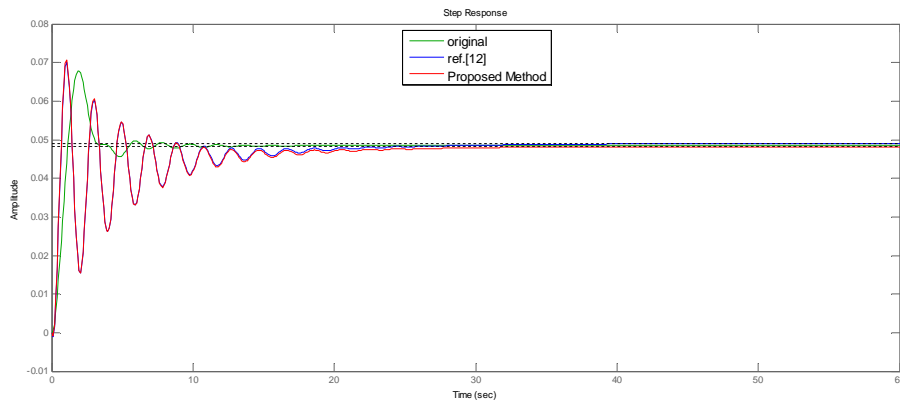


Fig 4.1 Comparison of Step responses of δ When $\Delta V_{Ref}(s) = 0.05$ p.u. and $\Delta T_m(s) = 0$.

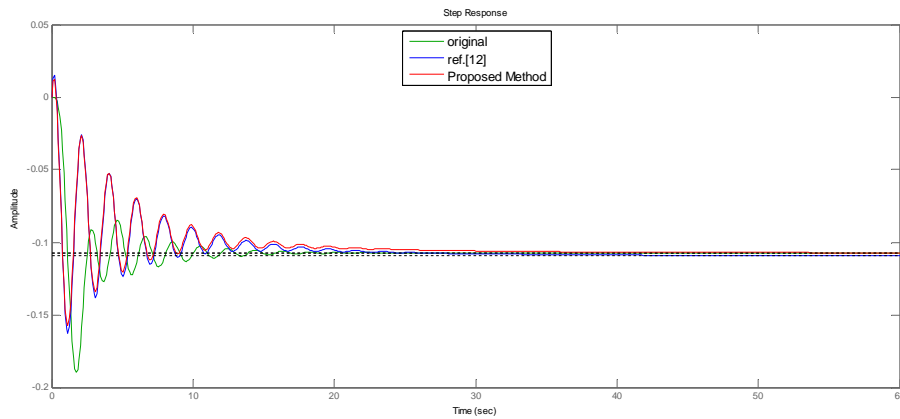


Fig 4.2 Comparison of Step responses of V_t When $\Delta V_{Ref}(s) = 0.05$ p.u. and $\Delta T_m(s) = 0$.

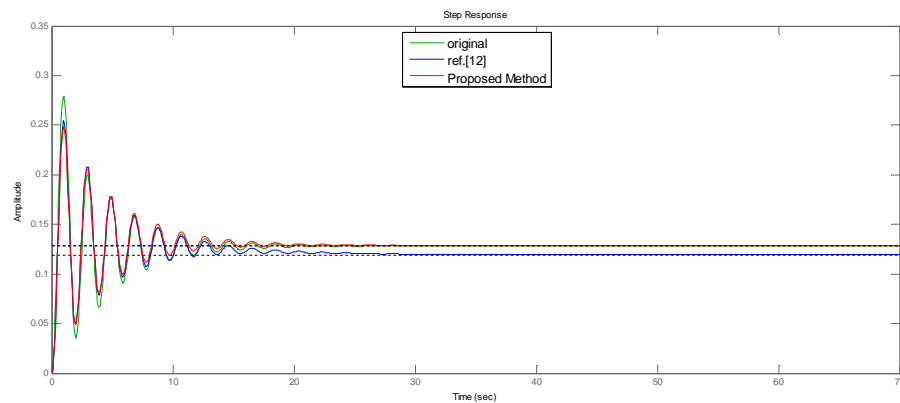


Fig 4.3 Comparison of Step responses of δ When $\Delta V_{Ref}(s) = 0$ and $\Delta T_m(s) = 0.05$ p.u.

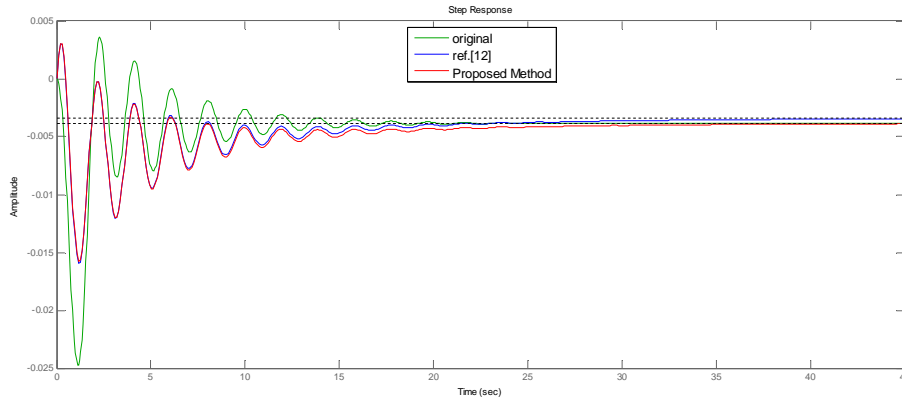


Fig4.4 Comparison of Step responses of V_t When $\Delta V_{Ref}(s) = 0$ and $\Delta T_m(s) = 0.05$ p.u.

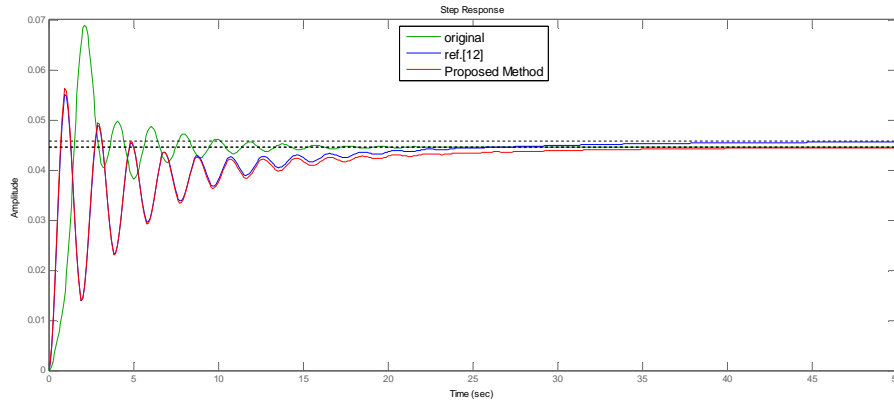


Fig 4.5 Comparison of Step responses of δ When $\Delta V_{Ref}(s) = 0.05$ p.u. and $\Delta T_m(s) = 0.05$ p.u.

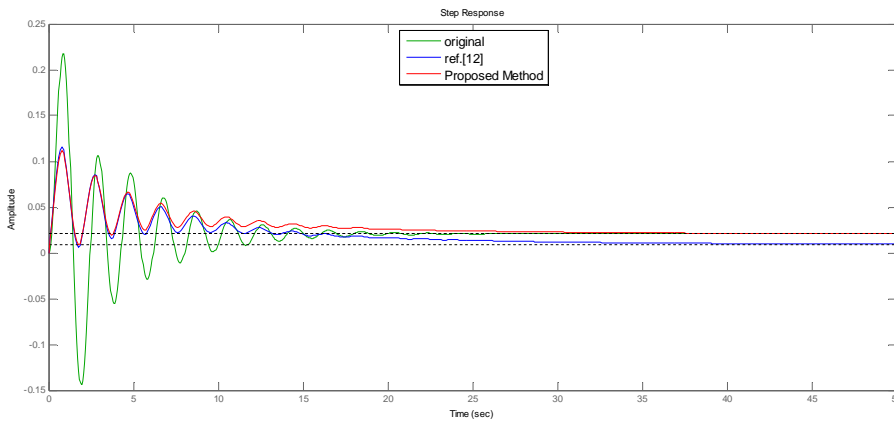


Fig4.6 Comparison of Step responses of V_t When $\Delta V_{Ref}(s) = 0.05$ p.u. and $\Delta T_m(s) = 0.05$ p.u.

From the responses obtained through the simulation it is clear that proposed method is closely matching with original system performance and is having a less settling time with stability being preserved for step variations in the input.

Table 4.1 Comparison of ISE of various reduced 3rd order models with their execution time.

Considered transfer function.	Proposed method (ISE)	Execution time of Proposed Method (sec)	G.Parmar [12] (ISE)	Execution time of [12] (sec)
$b_{11}(s)/D(s)$	8.7174	34.7905	8.96213	66.3999
$b_{12}(s)/D(s)$	0.5527	34.0691	3.03754	66.8405
$b_{21}(s)/D(s)$	1.0754	34.6958	1.19436	66.7224
$b_{22}(s)/D(s)$	0.0448	35.3835	0.06474	66.1706

From the Table 4.1 it is clear that the ISE values of the LOMs obtained by the proposed method is minimized in comparison with other methods available in the literature and it has shown its fast convergence .

5. CONCLUSION

The method retains dominant pole which always generates a stable LOM for a stable HOS and also retains the dynamic performance.This method allows the numerator coefficients of the HOS as free parameter in the process of order reduction. Numerator polynomials are obtained by minimising ISE and matching the steady state response using FA having a dominance facet of fast convergence and easy pursuit.Each element of the transfer function matrix of the HOS is assessed individually.This method is simple and cogent. The cognency of the proposed method is tested on a practical 10th order SMIB system.The step response of the original system and lower order model is almost allied.

LIST OF SYMBOLS AND ABBREVIATIONS

- P, Q Synchronous machine (sync.m) activepower, reactivepower
- δ, ω, V_t Sync.m torqueangle, speed, terminal voltage.
- $K_1, K_2, K_3, K_4, K_5, K_6$ Heffron-Phillips constants.
- T_a, T_e, T_m Sync.m accelerating, electrical and mechanical torque.
- H Sync.m inertia constant
- R_e, X_e External system equivalent resistance and reactance.
- E_q, E_{fd}, T_{do} Voltage proportional to d-axis flux linkages, field voltage and Open circuit time constant

K_E, S_E, T_E	Self-excited field constant, Saturation function and exciter time constant.
K_A, T_A, V_R	Regulator gain, time constant and output voltage.
K_f, T_f	RF gain and time constant
K_R, T_R	Filter gain and time constant
K_0, T_0, V_s	Speed gain, reset time-lag constant and voltage output of PSS
T_1, T_2, T_3, T_4	Lead-Lag time constants of PSS
Δ	Step change of input
FA	Firefly Algorithm
HOS	Higher Order System
ISE	Integral Square Error
LOM	Lower Order Model
MIMO	Multi-Input Multi-Output
PSS	Power System Stabilizer
RF	Rate Feedback
SMIB	Single Machine Infinite Bus

Appendix-A

Synchronous machine:

3-phase, 160 MVA, pf = 0.894, $x_d = 1.7$, $x_q = 1.6$, $x_d' = 0.245$ p.u., $\tau_{d0}' = 5.9$,
 $H = 5.4$ s, $\omega_r = 314$ rad s⁻¹.

Type-I exciter:

$K_A = 50$, $K_E = -0.17$, $S_E = 0.95$, $K_F = 0.04$, $K_R = 1$, $K_0 = 1$, $\tau_A = 0.05$, $\tau_E = 0.95$,
 $\tau_F = 1.0$, $\tau_R = 0.05$, $\tau_0 = 10.0$, $\tau_1 = \tau_3 = 0.440$ s, $\tau_2 = \tau_4 = 0.092$ s.

External System:

$R_e = 0.02$, $X_e = 0.40$ p.u. (on 160 MVA base).

Operating point:

$P_0 = 1.0$, $Q_0 = 0.5$, $E_{FD0} = 2.5128$, $E_{q0} = 0.9986$, $V_{t0} = 1.0$, $T_{mo} = 1.0$ p.u.,

$\delta_0 = 1.1966$ rad,

$K_1 = 1.1330$, $K_2 = 1.3295$, $K_3 = 0.3072$, $K_4 = 1.8235$, $K_5 = -0.0433$, $K_6 = 0.4777$.

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